

# Lightweight Composite Repair System

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## Introduction

- **Industry Shift:** Modern aviation is moving toward a heavy reliance on composite materials
- **Current Repair Bottleneck:** Producing quality repair parts is labor and energy intensive, sometimes requiring large scale equipment such as autoclaves
- **Innovation:** Development of a universal, out of autoclave repair system design for localized applications
- **Key Technology:** Utilizes lightweight expandable materials and localized heating
- **Result:** Achieves autoclave-equivalent quality and performance

## Problem Statement

- **Project Context:** Multi-year project, building upon previous years research and development
- **Primary Objective:** Design, fabricate and validate a portable lightweight composite repair system
- **Performance Requirements:**
  - **Pressure Generation:** Capable of maintaining 90 psi using expandable foam
  - **Thermal/Duration Requirements:** Sustains a constant 350 °F during cure
- **Engineering Constraints:**
  - **Out of Autoclave (OOA):** Must perform repair without external pressure vessels
  - **Structural:** Requires managing high reaction loads with limited mounting options

## Component Selection

- **Enclosure:** NOMEX Fabric & Wire Mesh
- **Cables:** Galvanized Steel
- **Foam:** Expandable Foam
- **Heating Element:** Heat Blanket Controlled with Heatcon Controller and Hot Plate
- **Filler Material:** Silicon Intensifier
- **Pressure Measurement:** FujiFilm LLW Hi-Temp



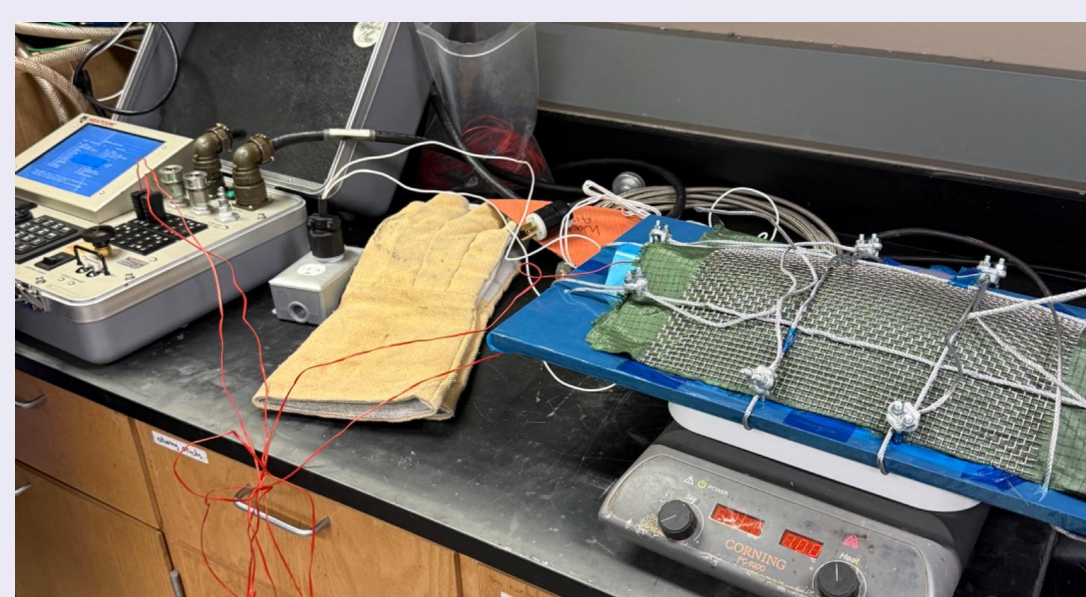
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### Fabricate Expandable foam

Refine expandable foam medium that expands and resists deformation under pressure to test with a mockup flat CFRP Plate

2026



### Lightweight Repair Apparatus

Design *lightweight* repair system using expandable medium and flexible cover and test on a mockup CFRP hat stringer

### Repair Housing Apparatus

Design and test a rigid housing apparatus that is capable of withstanding at least 100 psi at 350°F for at least an hour.



2025

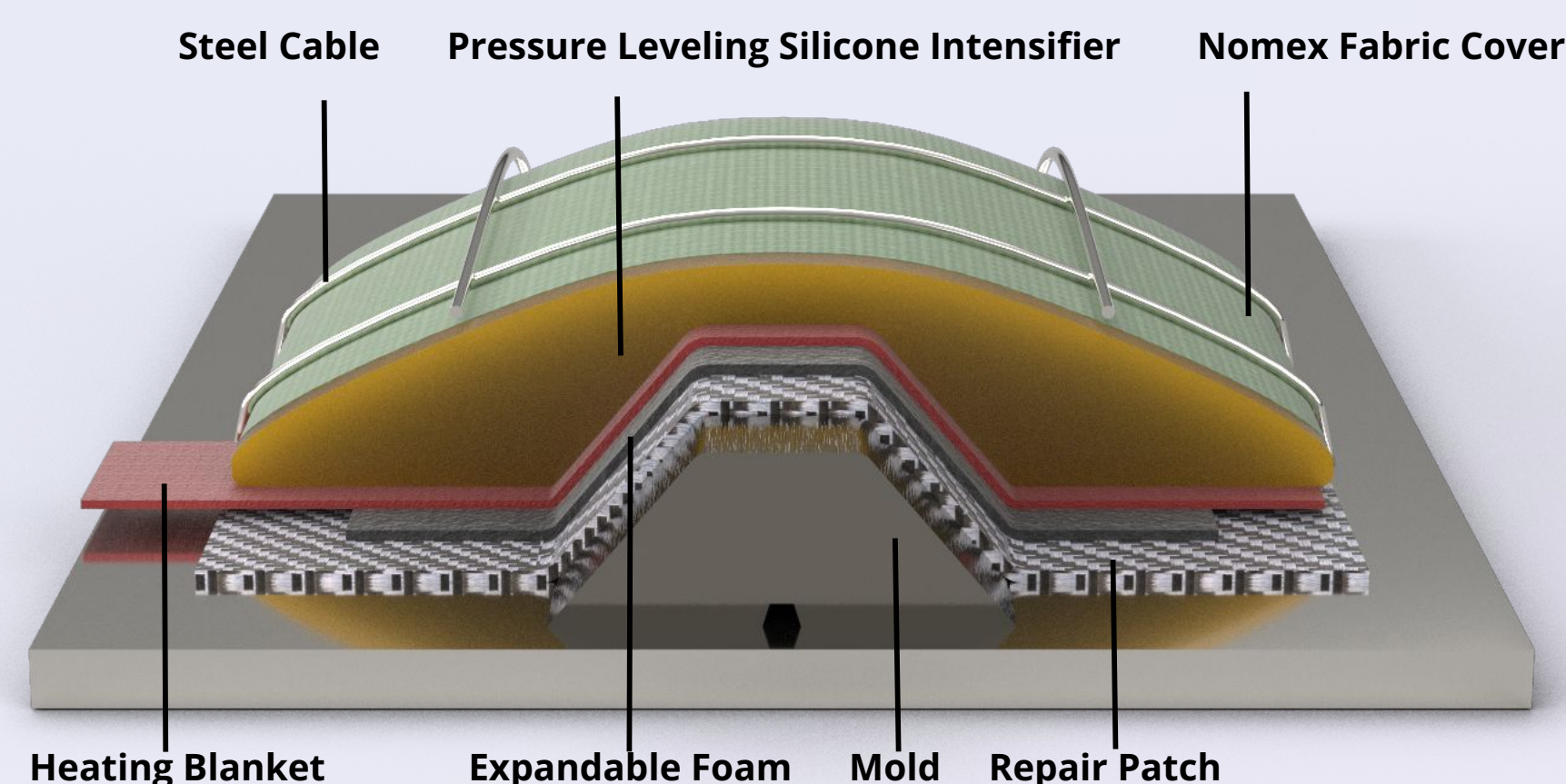
## Design

### Mounting Solution

- **Problem:** Generate reaction force caused by expandable medium without further damaging carbon fiber
- **Solution:** Steel cable shares load in tandem with Nomex fabric distributing reaction forces into switchable magnets

### Expandable Medium Solution

- **Problem:** Generate pressure and heat to cure autoclave specified CFRP parts for repairs
- **Solution:** Use proved expandable foam in tandem with silicon intensifier as a pressure leveling system



## Summary

### Optimal Results

- Optimal Outcomes: HB4 achieved highest tensile strength, HB5 highest cure degree, and HB3 lowest porosity.

### Reasoning

- HB5: Minimized thermal lag between top and bottom thermocouples
- HB5: Excessive pressure triggered resin leakage and micro-matrix cracking

### Pressure Map

- HB5: Maximum pressure: ≈ 300 psi. Minimum pressure ≈ 40 psi
- HB5: Severe pressure drop in the high curvature region between flange and ramp

### FEA

- Simulated maximum principal stress corroborates lack of pressure at flange to wall interface and concave corners while convex corners correspond to higher pressures
- Thermal simulation provides temperature gradients in repair region and equivalent thermal mapping for heat blanket effectiveness

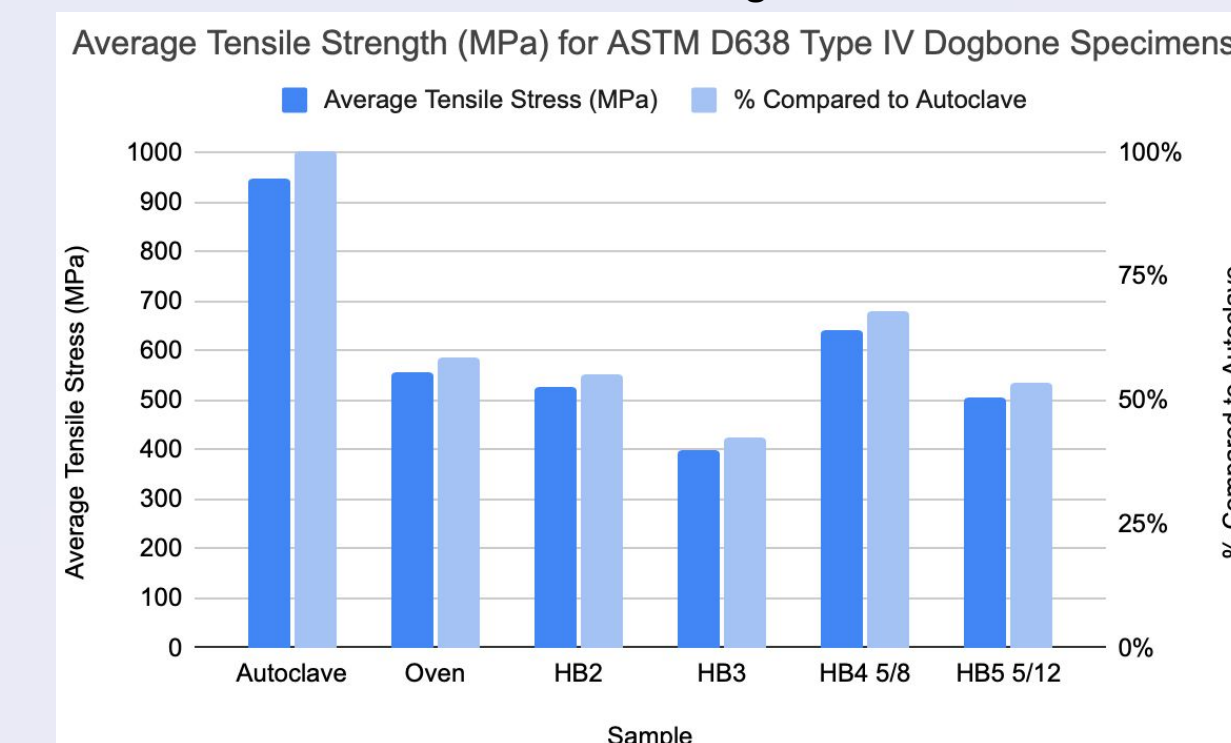
## Testing

### Differential Scanning Calorimetry: Degree of Cure

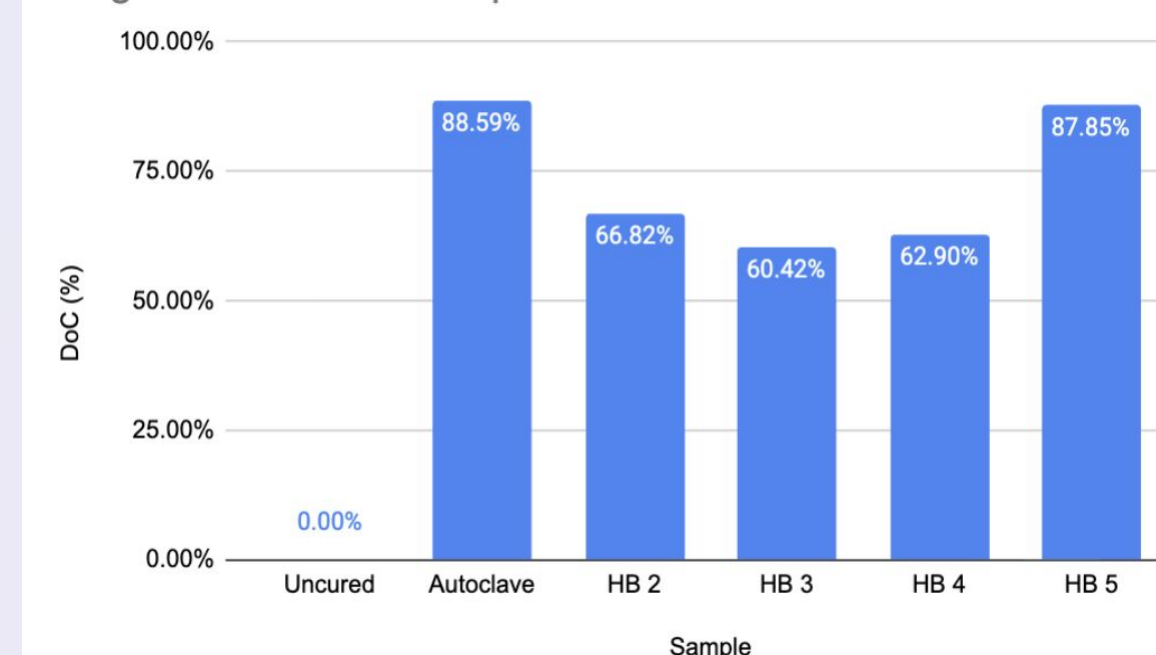
### Tensile: Ultimate Tensile Strength

### Microscopy: Porosity/Void %

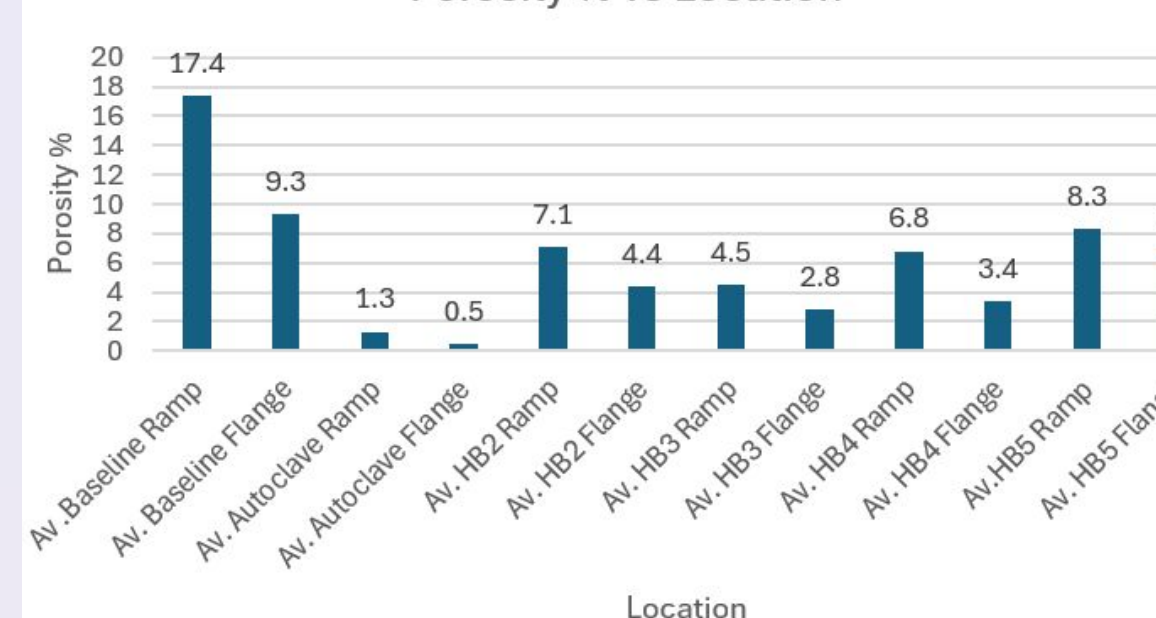
\*Samples for DSC are average of flange and ramp, tensile samples taken from flange



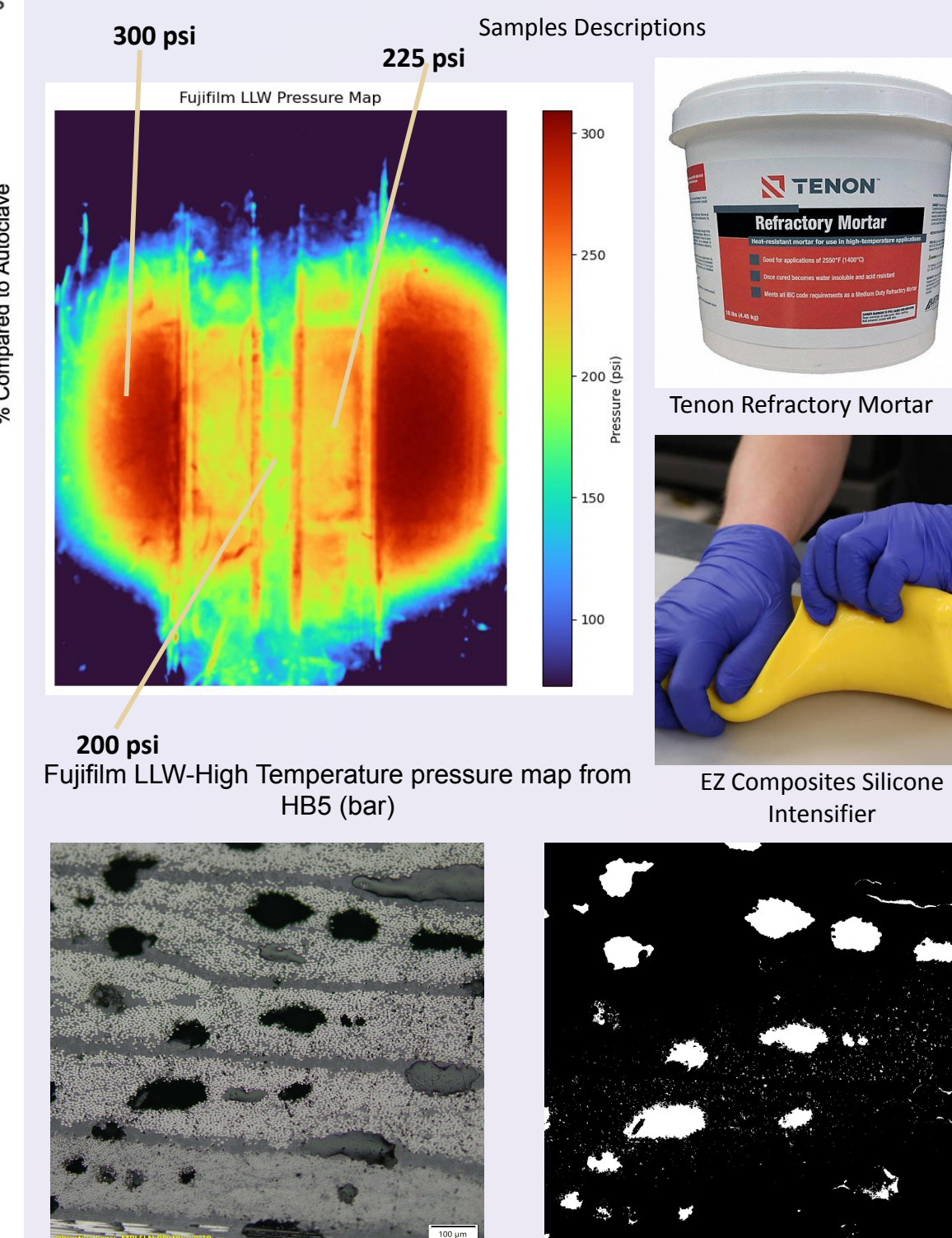
### Degree of Cure vs. Sample



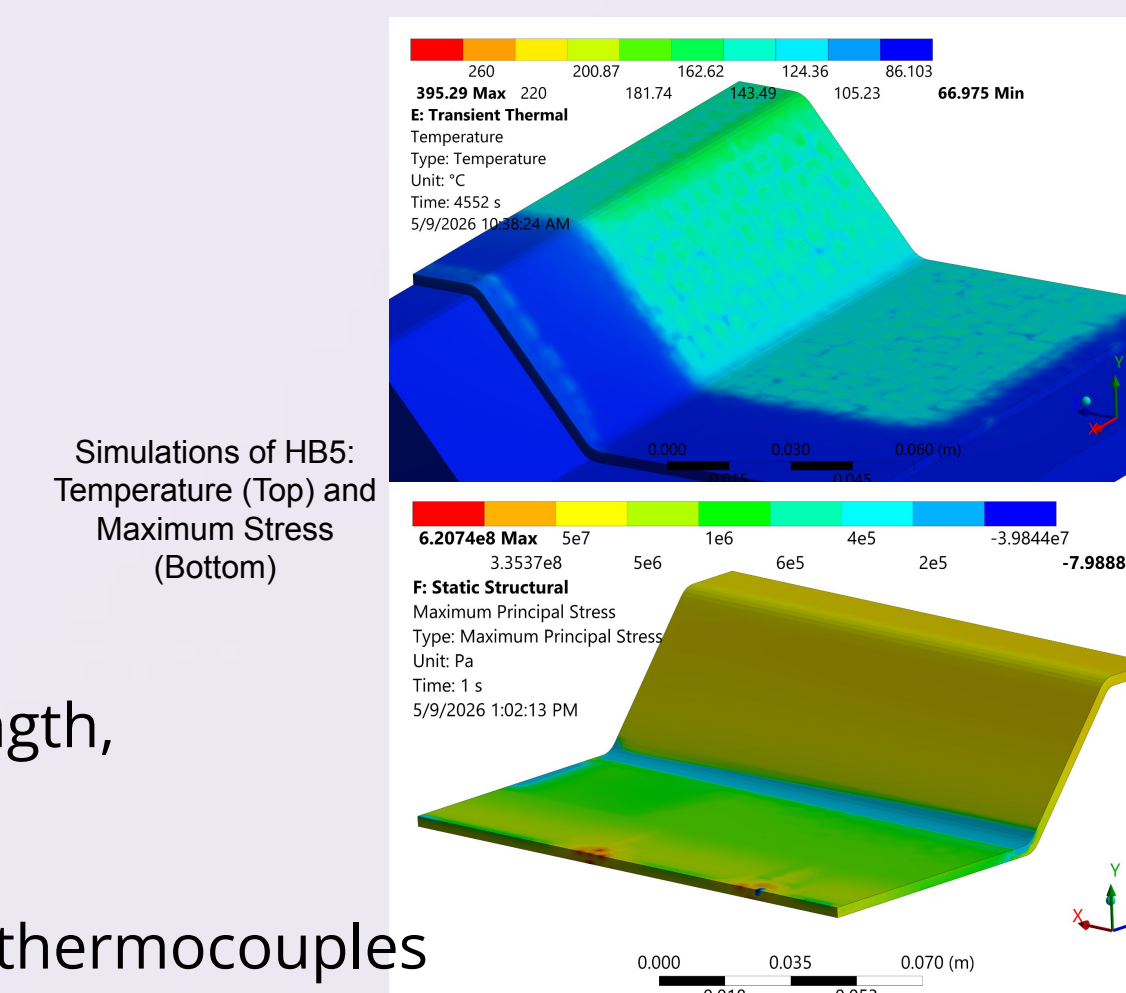
### Porosity % vs Location



Sample	Description
Autoclave	Standard Vacuum Bag
Oven	Clay with No Debulks
HB2	Clay with No Debulks
HB3	Clay + Intensifier with No Debulk
HB4	Only Intensifier with 4 Debulks
HB5	Only Intensifier with 5 Debulks



Raw microscopy (left) Processed image (right) Images are of HB5 Ramp



Simulations of HB5: Temperature (Top) and Maximum Stress (Bottom)